



SUPPLEMENTAL TYPE CERTIFICATE

10030229

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

STC TWENTY ONE LIMITED
The Waldorf Centre
1290 London Road
ALVASTON, DERBY
DE24 8QP
UNITED KINGDOM

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Product TC Number: FAA TCDS A2NM
TC Holder: BOEING
Model: B757-200/ -300

EASA Certification Basis:

The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate

As per FAA TCDS A2NM and as modified by CRI D-01 and CRI H-01

The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval

Description of Design Change:

Introduction of Air Conditioning Filter

Associated Technical Documentation:

STC 21 Compliance Record, S21.COMP-0694, Issue 1, 03/06/2010

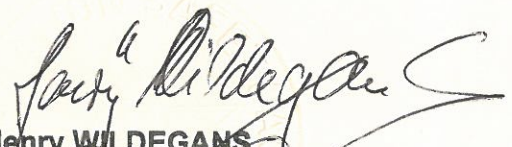
STC 21 Master Document List, S21.MDL-0694, Issue 1, 03/06/2010

or later revisions of the above listed documents approved by EASA

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 04.06.2010


Henry WILDEGANS
Project Certification Manager
Large Aeroplanes

Note:

The following numbers are listed on the certificate:

EASA current Project Number: 0010001189-001

EASA old Project Number: P-EASA.A.S.03972

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Limitations:

Installation of STC21 Limited Minor Change S21.21-15-1004, must be embodied prior to, or concurrent with, this STC

Conditions:

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product

This Certificate shall remain valid unless otherwise surrendered or revoked.

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